

Ice Prediction Workshop 2

Test Case Descriptions

IPW Organizing Committee

February 7, 2023

Revision log

- October 25, 2022
 - Initial version
- February 7, 2023
 - IRT cloud calibration data added for CRM65 cases simulation parameters
 - RG-15 AoAs changed to 4-degrees
 - RG-15 span changed from 590 to 580 mm
 - List of references

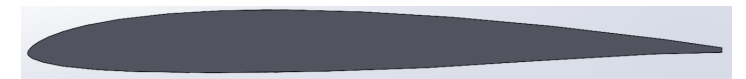
Configurations Summary



Case 1: CRM-65 Mid-span Hybrid (3D)



Case 2: CRM-65 Inboard Hybrid (3D)



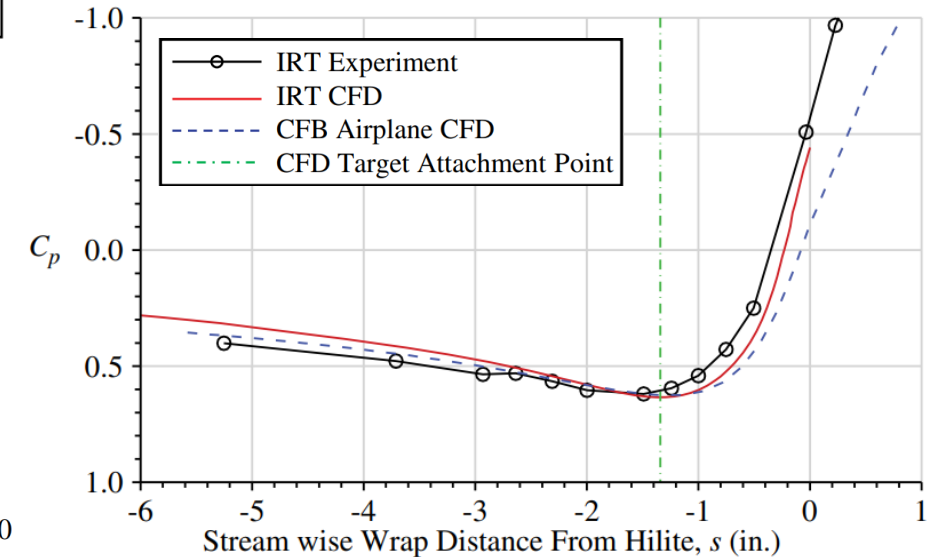
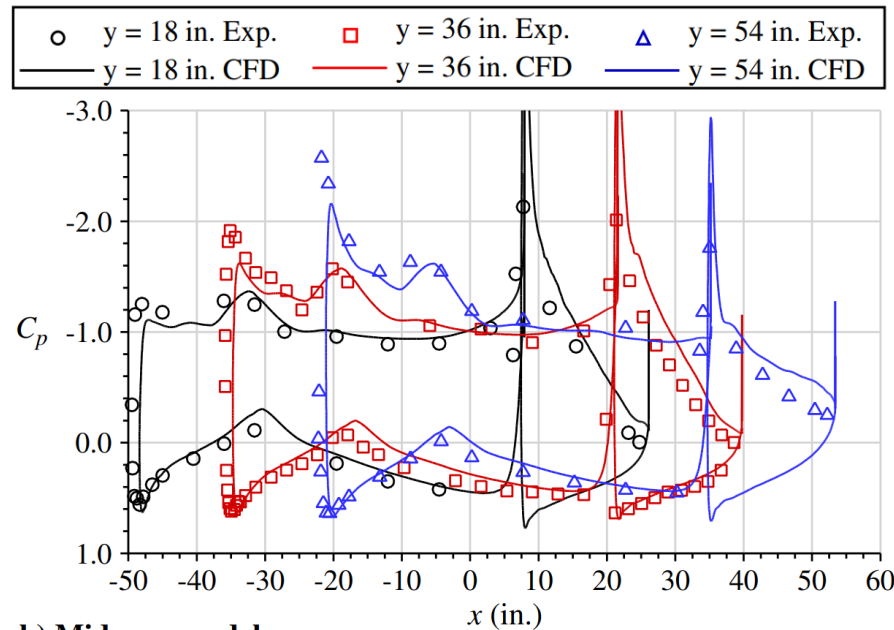
Case 3: RG-15 Low Speed Icing

Case list overview

IPW-2 Case no.	Configuration	AoA	Speed	T _{total} (°C)	LWC (g/m ³)	MVD (μm)	Icing Time (minutes)
1.1	CRM65 Mid-span	3.7	130 kts	-1.4	1.0	25	29
1.2	CRM65 Mid-span	3.7	130 kts	-6.3	1.0	25	29
1.3	CRM65 Mid-span	3.7	130 kts	-23.8	1.0	25	29
2.1	CRM65 Inboard	3.7	130 kts	-1.4	1.0	25	29
2.2	CRM65 Inboard	3.7	130 kts	-6.3	1.0	25	29
2.3	CRM65 Inboard	3.7	130 kts	-23.8	1.0	25	29
3.1	RG-15 Small wing	4	25 m/s	-2	0.44	24	20
3.2	RG-15 Small wing	4	25 m/s	-4	0.44	24	20
3.3	RG-15 Small wing	4	25 m/s	-10	0.44	24	20

Cases 1.1-1.3: CRM65 Midspan

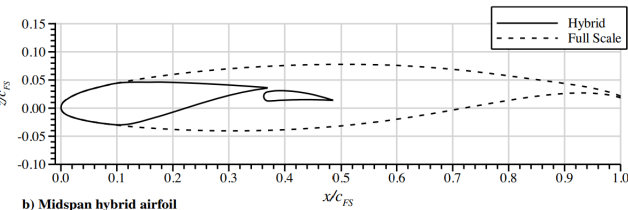
IPW-2 Case no	AoA (deg.)	Flap angle (deg.)	Speed (knots)	Speed (m/s)	T_{total} (°C)	T_{static} (°C)	P_{total} (kPa)	P_{static} (kPa)	MVD (μ m)	LWC (g/m ³)	Time (min.)	Freezing fraction	2015 ice mass (kg/m)	2021 ice mass (kg/m)	2022 ice mass (kg/m)
1.1	3.7	25.0	130	66.9	-1.4	-3.6	99.3	96.5	25	1.0	29.0	0.12	4.17	3.75	3.70
1.2	3.7	25.0	130	66.9	-6.3	-8.5	97.5	94.7	25	1.0	29.0	0.35	6.42	6.64	5.93
1.3	3.7	25.0	130	66.9	-23.8	-26.0	99.3	96.3	25	1.0	29.0	1.00	6.5	5.43	5.30



b) Midspan model

b) Midspan model

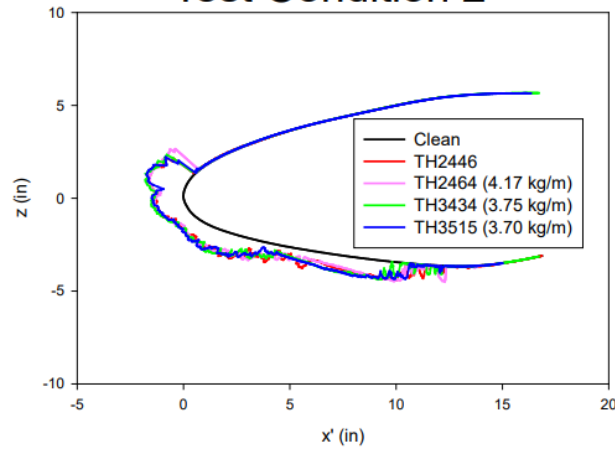
- Workshop CAD and grids are provided with AoA and flap angle built-in
- Pressure coefficients are based on inlet static pressure and density: CFD results that set back pressure at tunnel exit should use converged inlet static pressure as reference when computing C_p .



Cases 1.1-1.3: CRM65 Mid-span: Comparison of MCCS

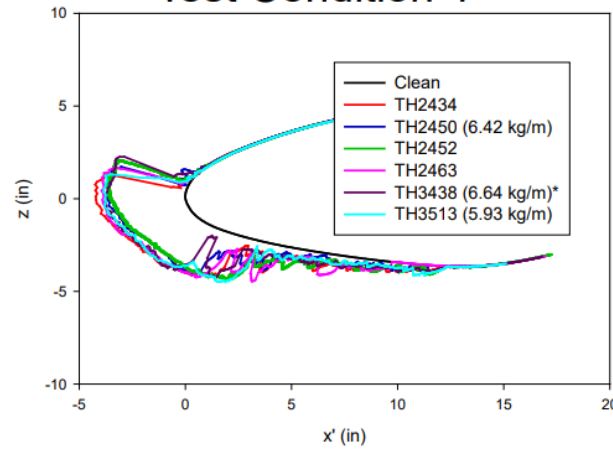
IPW-2 Case no	AoA (deg.)	Flap angle (deg.)	Speed (knots)	Speed (m/s)	T _{total} (°C)	T _{static} (°C)	P _{total} (kPa)	P _{static} (kPa)	MVD (μm)	LWC (g/m ³)	Time (min.)	Freezing fraction	2015 ice mass (kg/m)	2021 ice mass (kg/m)	2022 ice mass (kg/m)
1.1	3.7	25.0	130	66.9	-1.4	-3.6	99.3	96.5	25	1.0	29.0	0.12	4.17	3.75	3.70
1.2	3.7	25.0	130	66.9	-6.3	-8.5	97.5	94.7	25	1.0	29.0	0.35	6.42	6.64	5.93
1.3	3.7	25.0	130	66.9	-23.8	-26.0	99.3	96.3	25	1.0	29.0	1.00	6.5	5.43	5.30

Test Condition 2



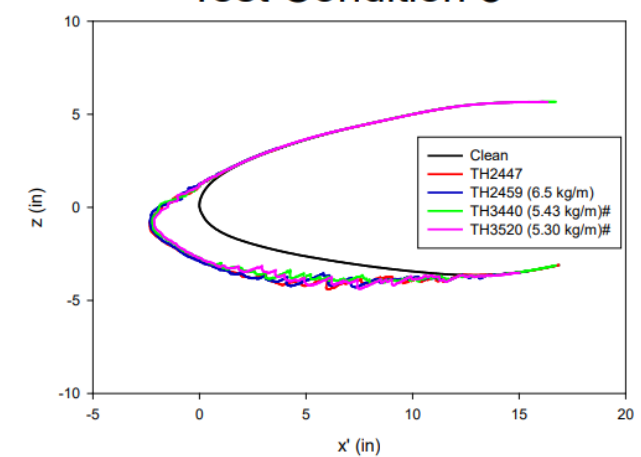
Case 1.1

Test Condition 4



Case 1.2

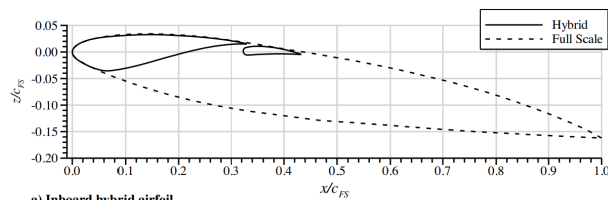
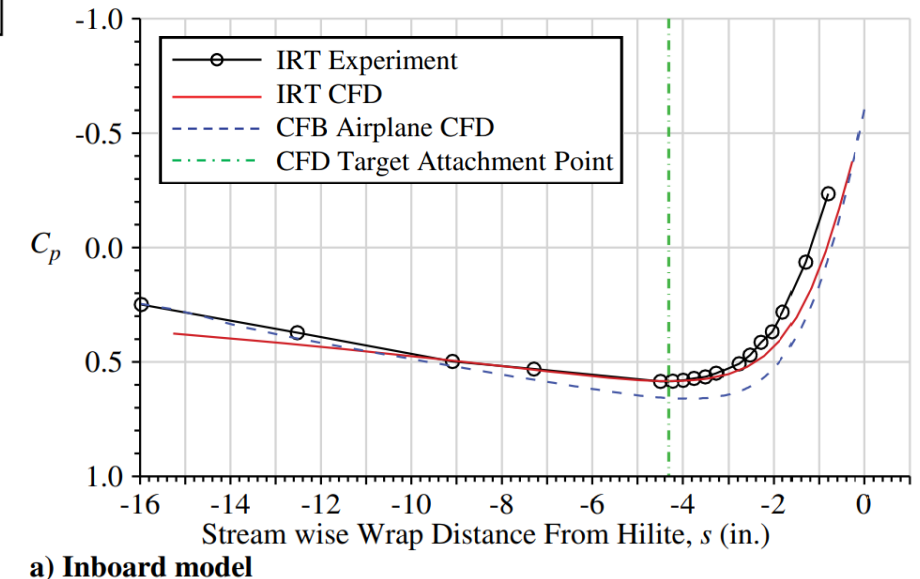
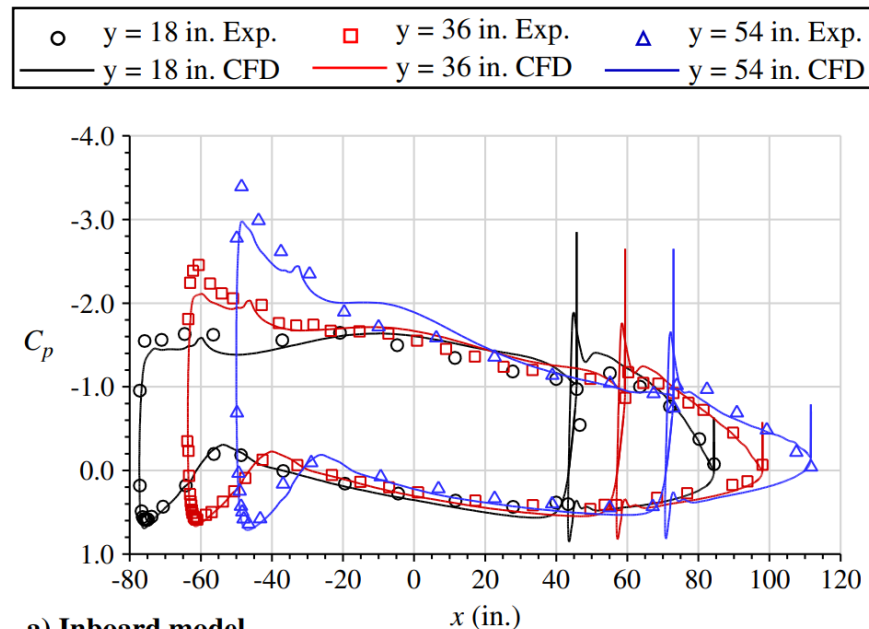
Test Condition 8



Case 1.3

Cases 2.1-2.3: CRM65 Inboard

IPW-2 Case no	AoA (deg.)	Flap angle (deg.)	Speed (knots)	Speed (m/s)	T_{total} (°C)	T_{static} (°C)	P_{total} (kPa)	P_{static} (kPa)	MVD (μm)	LWC (g/m ³)	Time (min.)	Freezing fraction	2015 ice mass (kg/m)
2.1	3.7	13.8	130	66.9	-1.4	-3.6	99.3	96.5	25	1.0	29.0	0.12	4.92
2.2	3.7	13.8	130	66.9	-6.3	-8.5	100.7	97.8	25	1.0	29.0	0.35	8.22
2.3	3.7	13.8	130	66.9	-23.8	-26.0	99.3	96.3	25	1.0	29.0	1.00	7.90

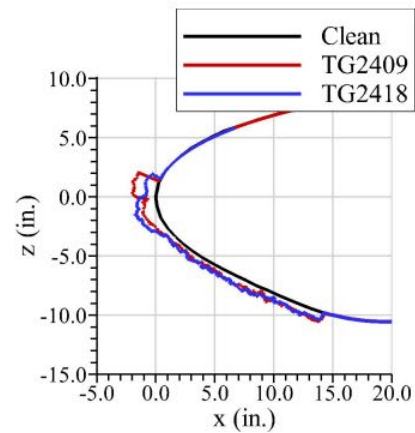


- Workshop CAD and grids are provided with AoA and flap angle built-in
- Pressure coefficients are based on inlet static pressure and density: CFD results that set back pressure at tunnel exit should use converged inlet static pressure as reference when computing C_p .

Cases 2.1-2.3: CRM65 Inboard: Comparison of MCCS

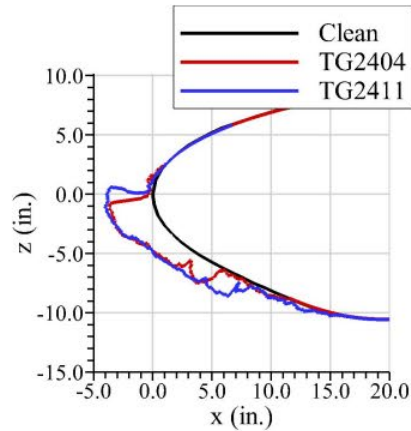
IPW-2 Case no	AoA (deg.)	Flap angle (deg.)	Speed (knots)	Speed (m/s)	T _{total} (°C)	T _{static} (°C)	P _{total} (kPa)	P _{static} (kPa)	MVD (μm)	LWC (g/m ³)	Time (min.)	Freezing fraction	2015 ice mass (kg/m)
2.1	3.7	13.8	130	66.9	-1.4	-3.6	99.3	96.5	25	1.0	29.0	0.12	4.92
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2.3	3.7	13.8	130	66.9	-23.8	-26.0	99.3	96.3	25	1.0	29.0	1.00	7.90

Test Condition 2



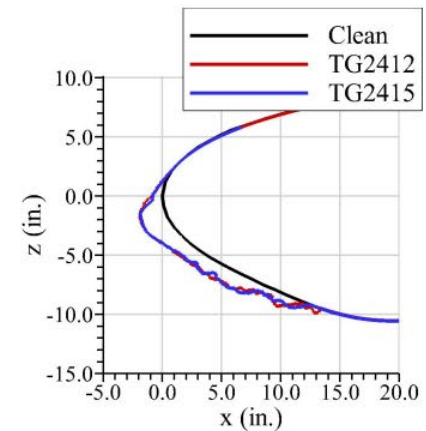
Case 2.1

Test Condition 4



Case 2.2

Test Condition 8



Case 2.3

NASA Glenn Research Center Icing Research Tunnel Cloud Calibration data

- This data is provided as recommendation for setting numerical simulation parameters
- Below is the cloud uniformity map for MVD = 20 μm and V = 150 knots take from NASA TM 2015-218758. The contour levels are ice thickness on the grid normalized by the average of the center 12 locations. Note that the test section is 72-inches tall.

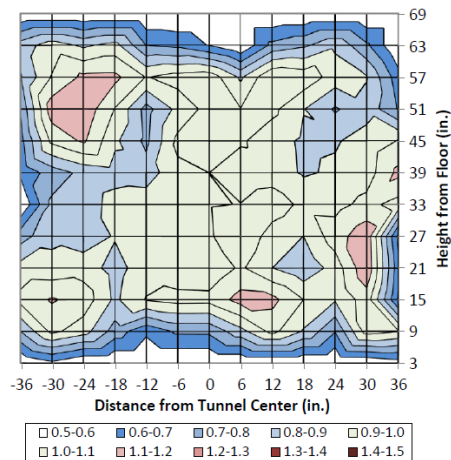


Figure 2.—A technician measures the thickness of ice accreted on the Grid.

- IRT distribution is different than Langmuir-D, the table below lists the recommended 7-bin distribution

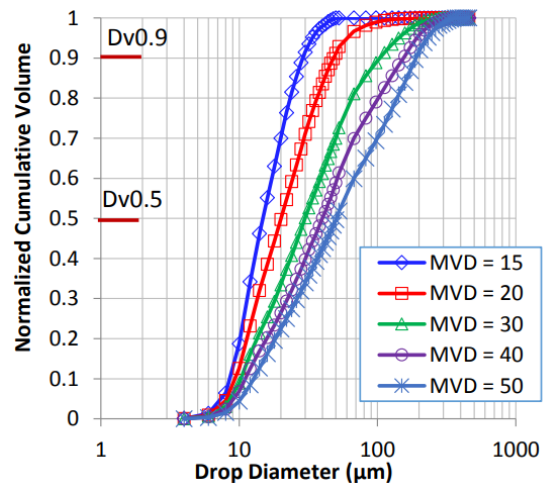
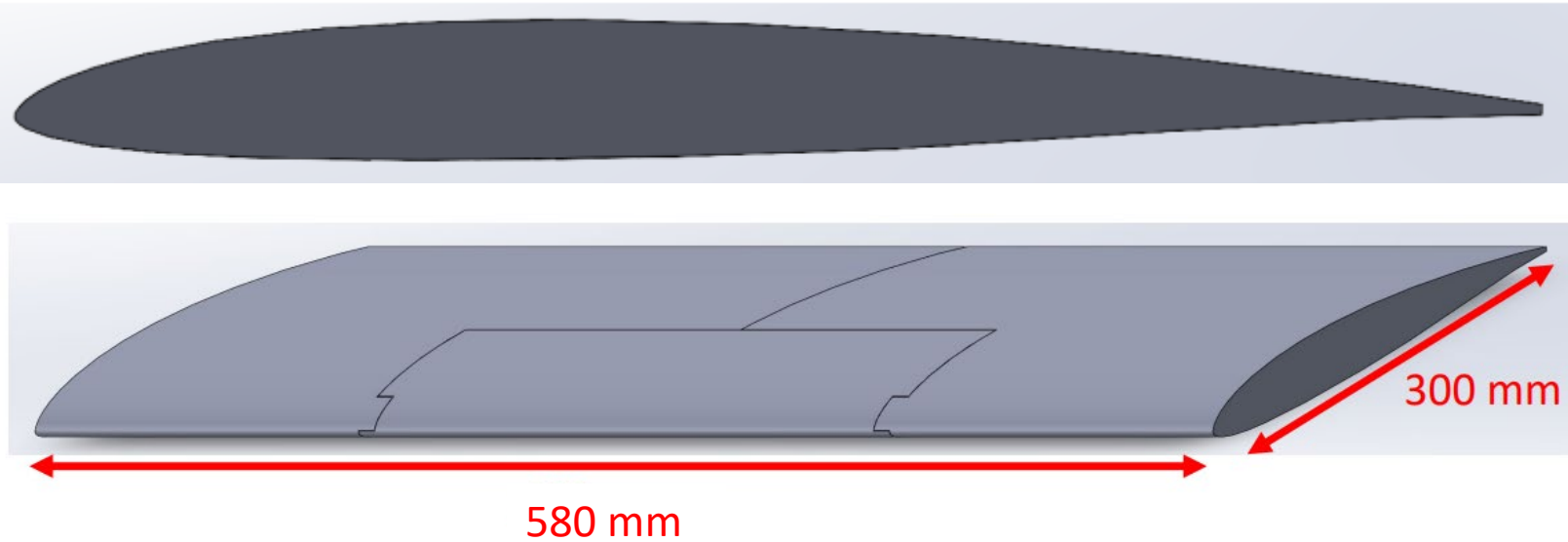
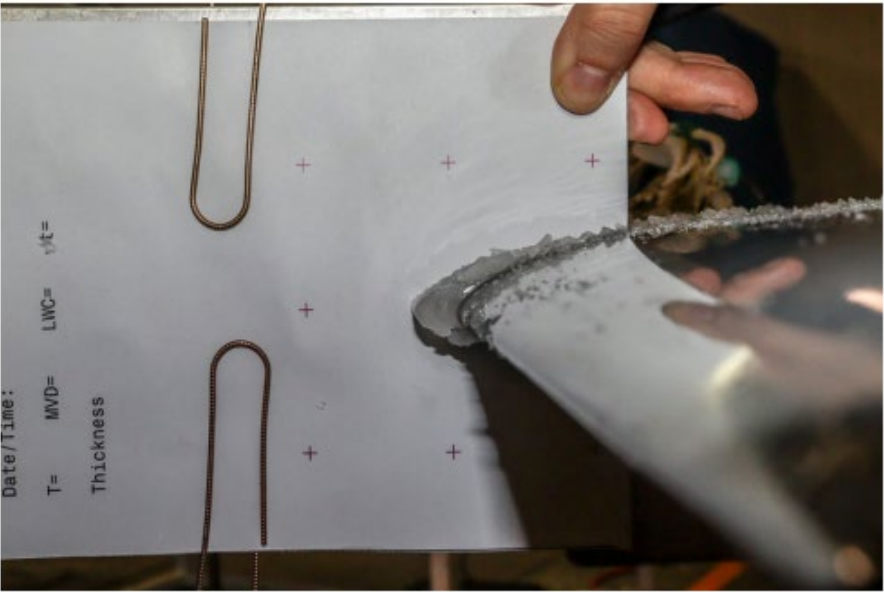


TABLE 4.1.—IRT 2015 7-BIN DROP DISTRIBUTIONS				
LWC, percent	Cumulative LWC, percent	Drop sizes with MVD = 20 μm , μm	Drop sizes with MVD = 50 μm , μm	Drop sizes linearly interpolated for MVD = 25 μm , μm
5	5	6.8	8.2	7.0
10	15	9.3	11.7	9.7
20	35	12.3	21.9	13.9
30	65	19.6	50.9	24.8
20	85	31.8	114.6	45.6
10	95	47.7	183.7	70.4
5	100	78.1	258.5	108.2

Cases 3.1-3.3: RG-15 Small Wing Low Speed Icing

IPW-2 Case no.	AoA (deg.)	Speed (m/s)	T _{static} (°C)	P _{static} (kPa)	MVD (μm)	LWC (g/m³)	Time (min.)
3.1	4	25	-2.0	101.3	24	0.44	20
3.2	4	25	-4.0	101.3	24	0.44	20
3.3	4	25	-10.0	101.3	24	0.44	20

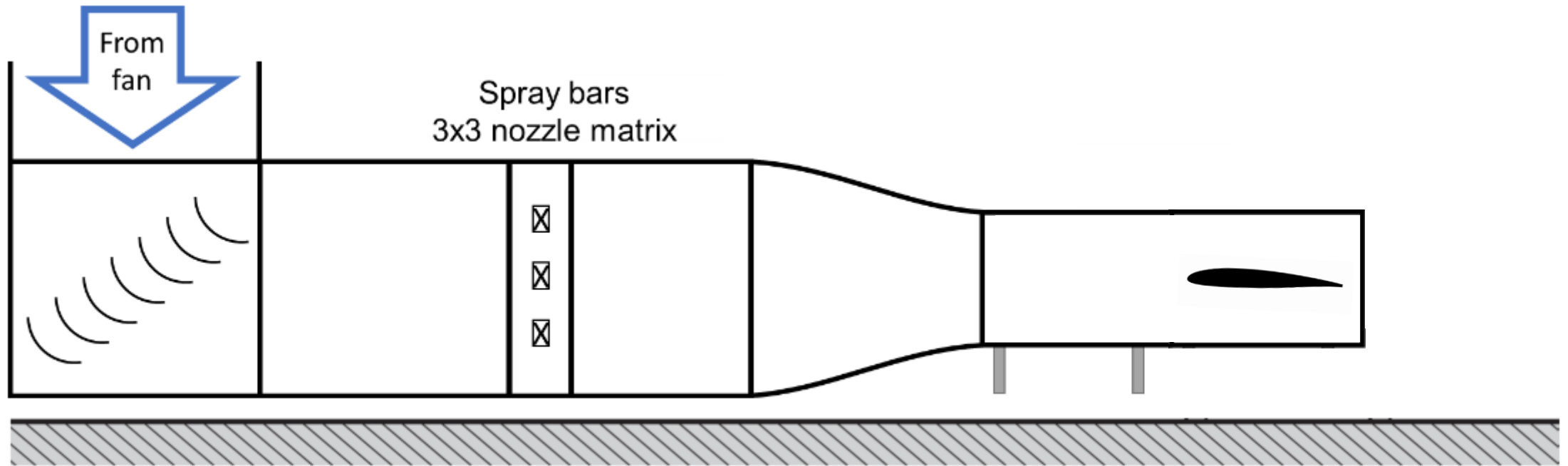


Cases 3.1-3.3: RG-15 Small Wing Low Speed Icing

Airfoil	RG-15
Span	0.58 m
Chord	0.30 m
Airspeed	25 m/s
Angle of attack	4 °
Liquid water content (LWC)	0.44 g/m ³
Mean volume diameter (MVD)	24 microns*
Duration	20 min
Static temperature (glaze, mixed, rime)	[−2, −4, −10] °C
Reynolds numbers	[5.7, 5.8, 6.0]×10 ⁵
Relative Humidity	95–100%
Pressure	101.3 kPa

*Monodisperse. A more detailed droplet spectrum will be supplied at a later stage.

Icing wind tunnel setup

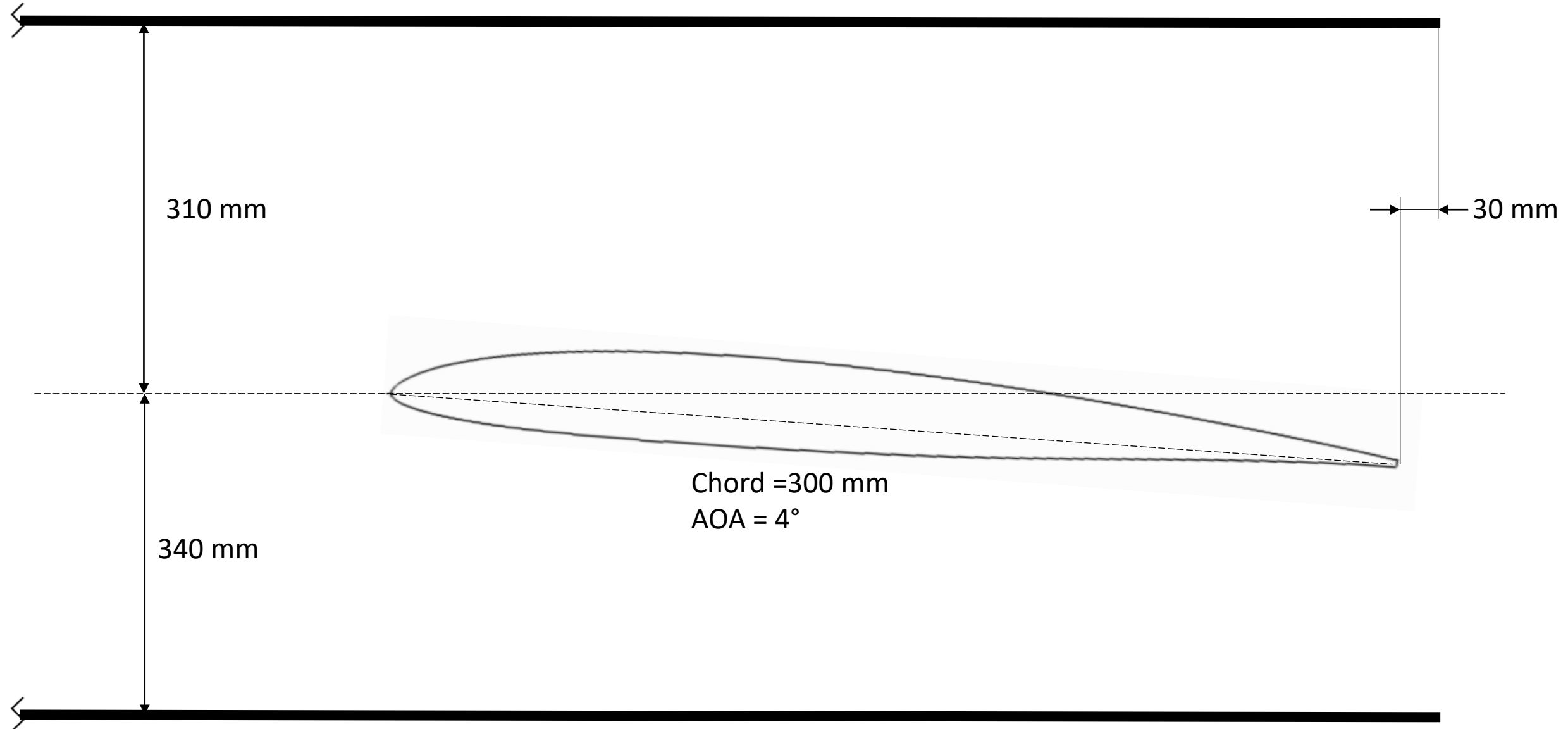


Cases 3.1-3.3: RG-15 Small Wing Low Speed Icing



Cases 3.1-3.3: RG-15 Small Wing Low Speed Icing

Tunnel cross-section



Mounting plate, see next slide

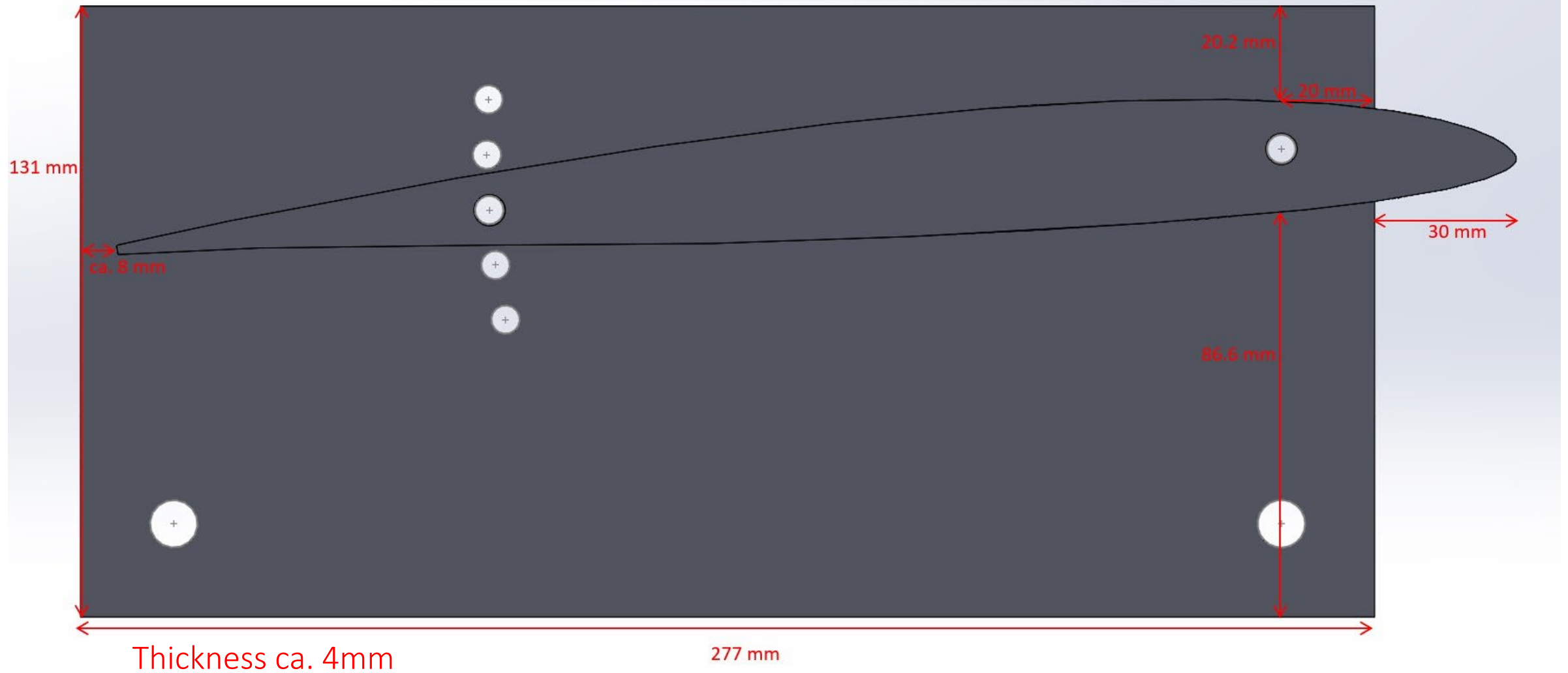
60 mm

580 mm

60 mm

Cases 3.1-3.3: RG-15 Small Wing Low Speed Icing

Mounting plates



References

- Broeren, A., Potapczuk, M., Lee, S., Malone, A., Paul, B., Woodard, B. “Ice-Accretion Test Results for Three Large-Scale Swept-Wing Models in the NASA Icing Research Tunnel”, AIAA-2016-3733, doi: 10.2514/6.2016-3733
- Yadlin, Y., Monnig, J., Malone, A., Paul, B., “Icing Simulation Research Supporting the Ice-Accretion Testing of Large-Scale Swept-Wing Models”, NASA/CR—2018-219781
- Fujiwara, G., Bragg, M., Broeren, A., “Comparison of Computational and Experimental Ice Accretions of Large Swept Wings”, Journal of Aircraft Vol. 57, No. 2, March–April 2020
- Steen, L., Ide, R., Zante, J., Acosta, W., “NASA Glenn Icing Research Tunnel: 2014 and 2015 Cloud Calibration Procedures and Results”, NASA/TM—2015-218758
- Broeren, A., Lee, S., Bragg, M., Woodard, B., Radenac, E., Moens, F., “Experimental and Computational Icing Simulation for Large Swept Wings”, NASA/TP-20210023843
- Hann, R., Müller, M., Lindner, M., Wallisch, J., UAV Icing: “Experimental validation data for predicting ice shapes at low Reynolds numbers”, Technical Paper, International Conference on Icing of Aircraft, Engines, and Structures, accepted manuscript, 2023.